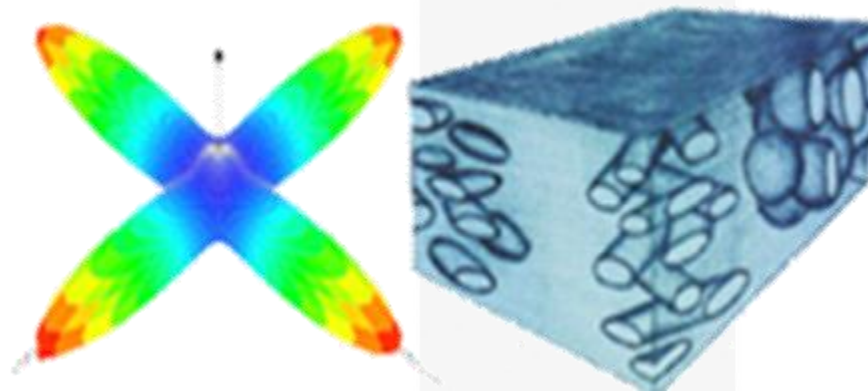


Material characterization of composites using micro mechanic models as key enabler

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4a engineering GmbH



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Industriepark 1
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reithofer@4a.co.at
++43 (0) 664 80106 601

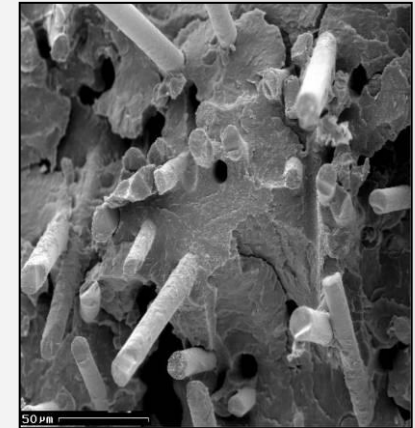
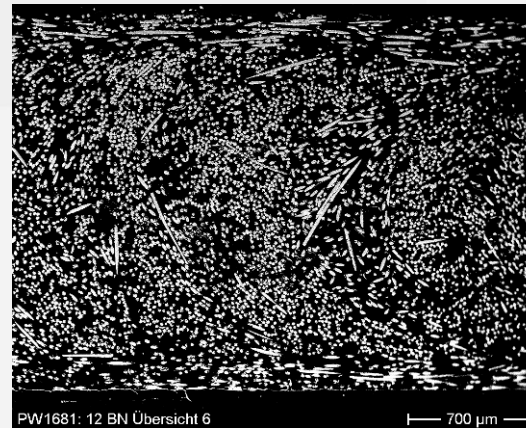


automotive CAE Grand Challenge 2016
April 12 - 13, 2016
Congress Park Hanau, Germany

- Introduction
- Exemplary material behavior
fiber orientation and content – strain rate - temperature
- *MAT_24 – typical approach
- Available material models in LS-DYNA
- Micro mechanic based model
- Simulation process chain
- Case studies
- Conclusion

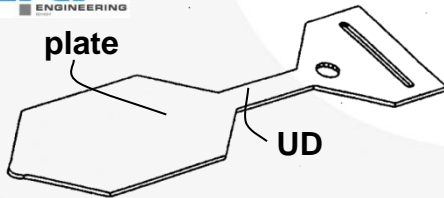
Fiber size and geometry → significant influence on part performance

Increasing fiber content → orthotropic properties increase & effect of strain rate diminishes (less content of matrix material)



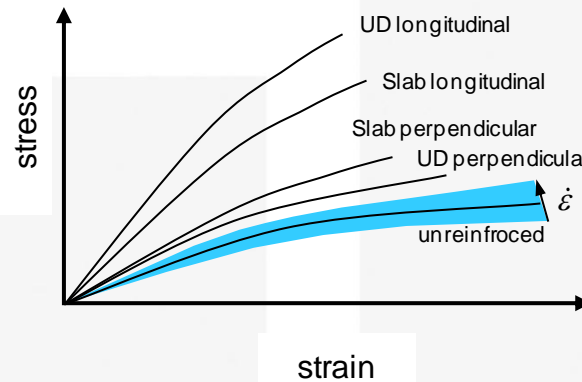
4a experimental tool

plate

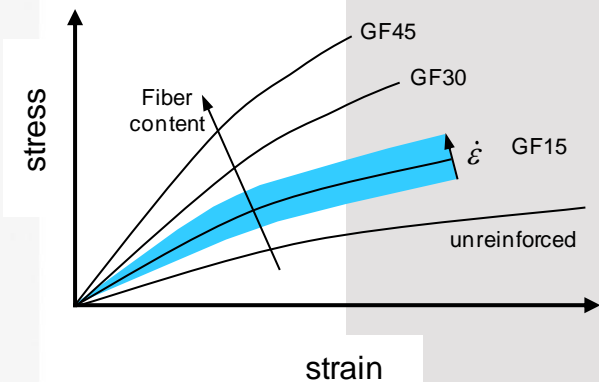


Fiber

dependence on fiber orientation



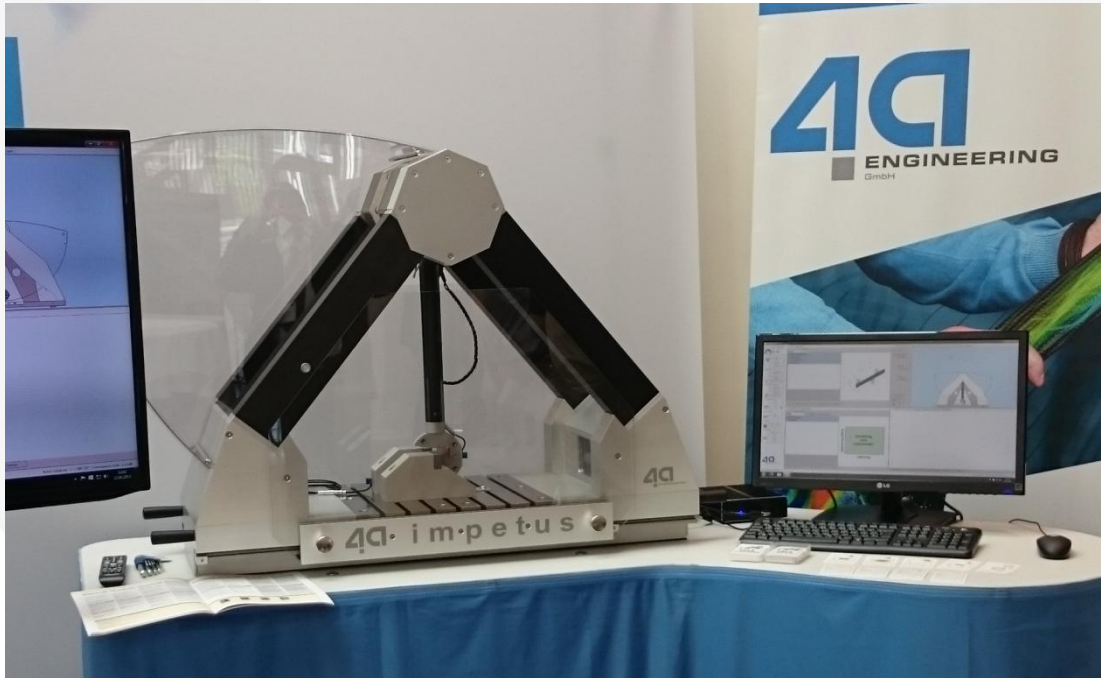
dependence on fiber content



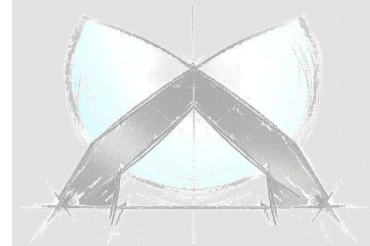
- efficient high-dynamic testing
- crash-behavior of plastics
- material data for simulation



bending test on 4a impetus

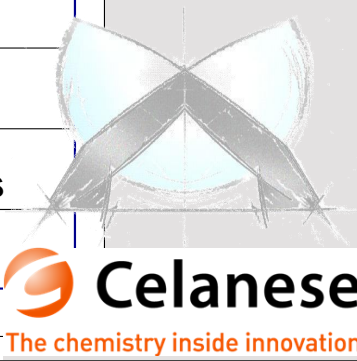
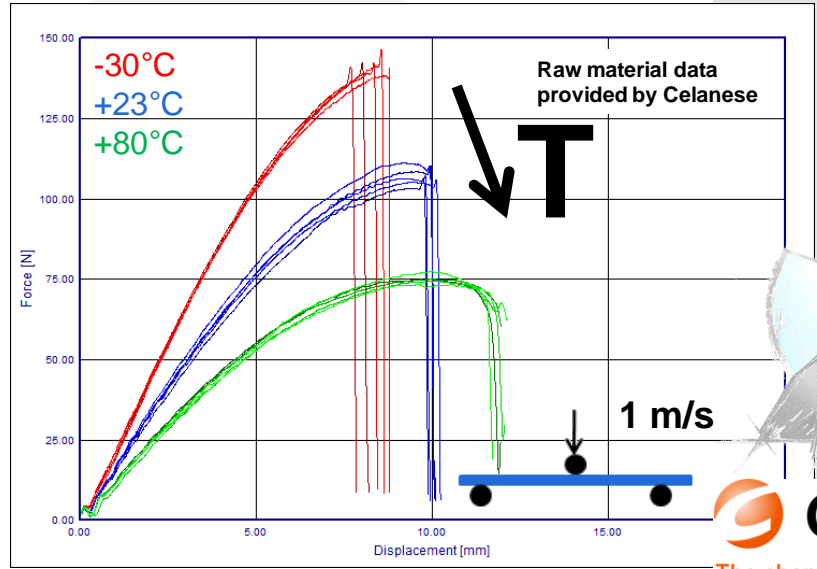
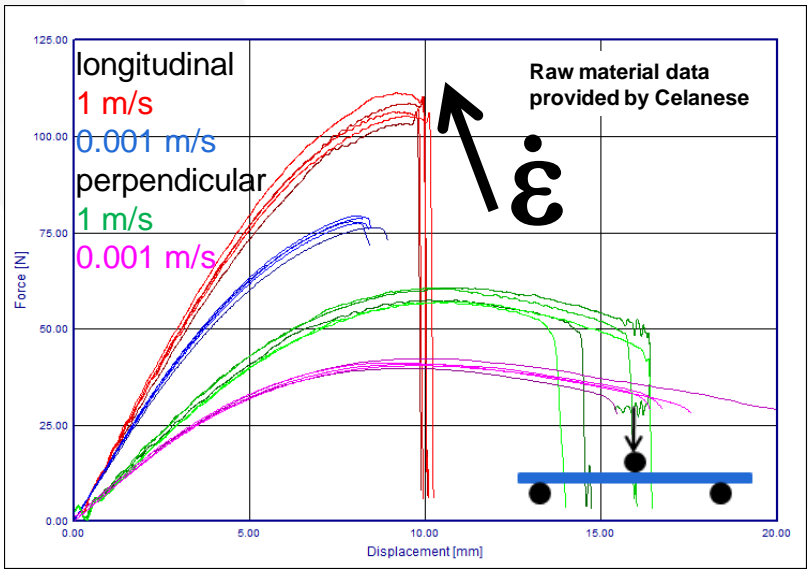
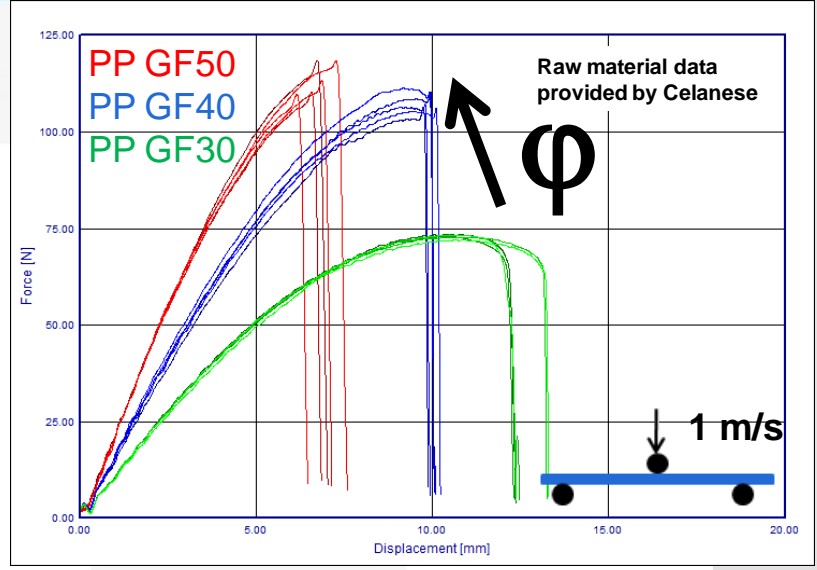
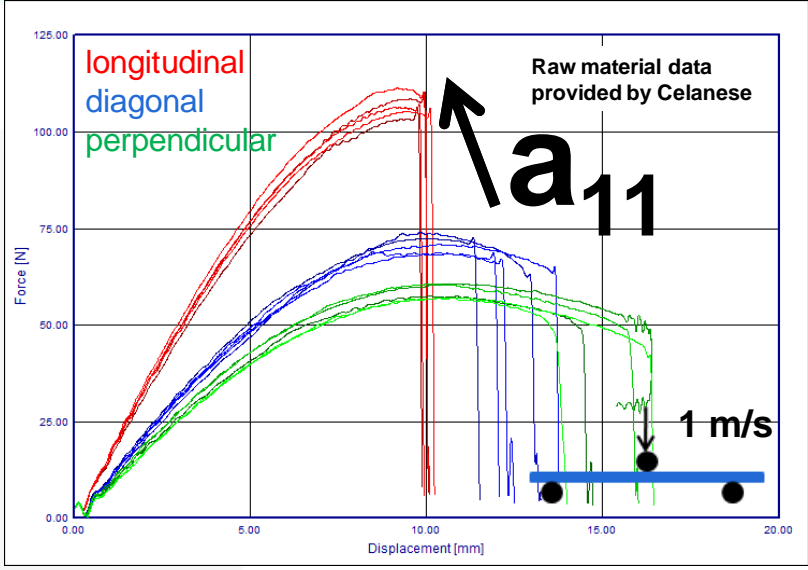


**4a impetus - intelligent testing systems
powered by 4a engineering GmbH**



Fiber reinforced thermoplastics

3-point-bending, support distance 50 mm



4a Impetus v3.00 beta

licensed for Peter Reithofer 4aengineering

Versuchsdaten: Neu, Kopieren, Prüfplan, Versuchsdurchführung, Set Winkel, Start, Report, Messkurven

Versuchsauswertung: Auswerten, Load Plots, Unload Plots, Mittelwertplots

Parametermodell: Neu, Kopieren, Hinzufügen

Optimierung: View, Start

Material: akt. Modell plotten, Karte erstellen

4a Impetus: Datenverzeichnis, Hilfe, Beenden

Versuch/Test: Database Tests, Datum_Nr: 110125_001

Allgemeine Informationen

Prüfsetup

Probekörper

Auswertung

Filter	0	kein Filter
Geschwindigkeitsermittlung	3 - Weg (statisch Anfangsanstieg)	
Nullpunktbestimmung	0 - manuelle (Zeitpunkt)	
Brucherkennung	0 - manuell (Zeitpunkt)	
Spannungsauswertung		
Steifigkeitsauswertung		
Ergebnisse		

3D Model Dimensions: L, LW, D, t

Graph: Channels [I], Externe Daten, Externe Daten

Parametermodell: Modelle, Datum_Nr_Name: 110501_001, Material, Designvariable

Modellereinstellungen

Werkstoff

Materialklasse	
Materialtyp	Thermoplast unv.

Materialverhalten

Materialquelle	benutzerdefiniert
Materialkarte	339

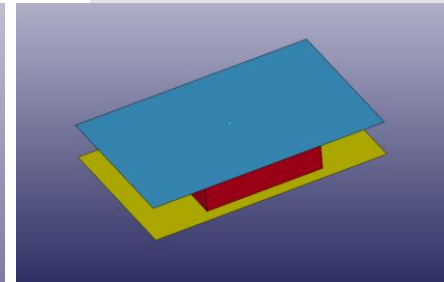
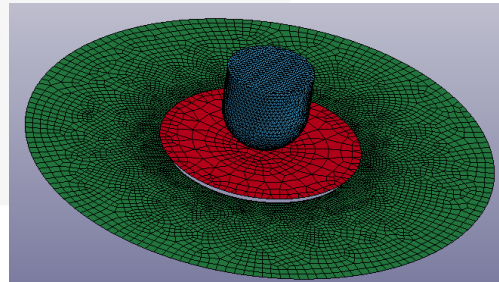
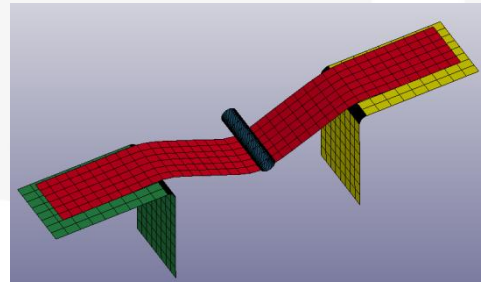
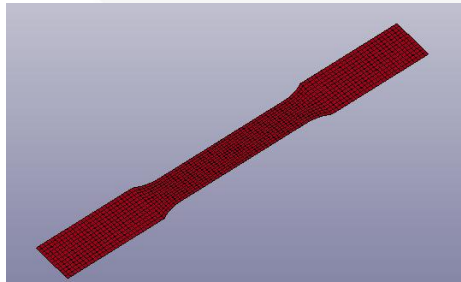
Idealisierung

Einheitensystem	SI(kg-m-sec-Pa)
Solver	LS DYNA
Inputdeck	customized
Modellsymmetrie	Viertelmodell
Elementtyp	Solid Hexaeder
Elementgröße	2
Kontaktstärke	0
Elementschichten	0
Benutzerdefinierter Parameter	0

Graph: Kraft [N] vs. Weg [mm]

Active Datenbank: L:\UA\EN\Projekte\11052101_hostacom\main_v3.mdb

**Complete system
from the test to the validated
material card**

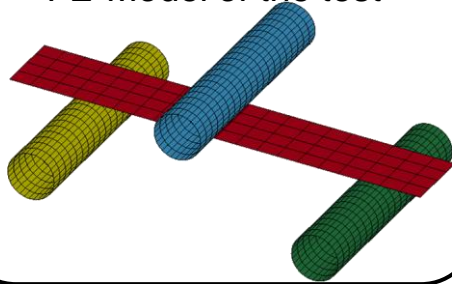


[2005Fritz]

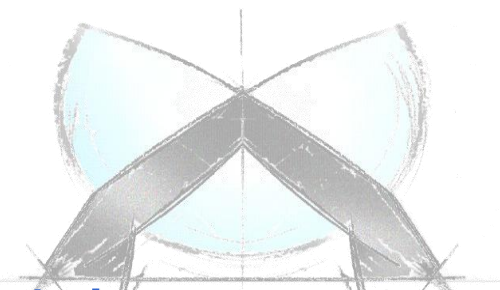
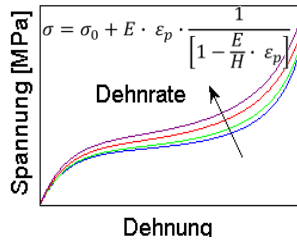
*MAT_24 – typical approach

Separate for each direction

FE-Model of the test



Parameterized materialcard



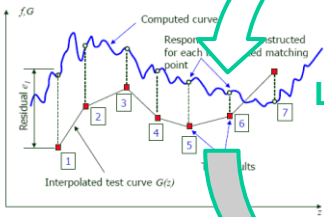
4a impetus

© by 4a engineering GmbH - intelligent testing systems

Hyperview, Animator, LS PREPOST, ...

DATABASE
measurement, models

$$\epsilon = \frac{1}{P} \sum_{i=1}^n W_i \left(\frac{f_i(x) - G_f}{\epsilon_p} \right)^2 = \frac{1}{P} \sum_{i=1}^n W_i \left(\frac{\sigma_i(x)}{\sigma_p} \right)^2$$

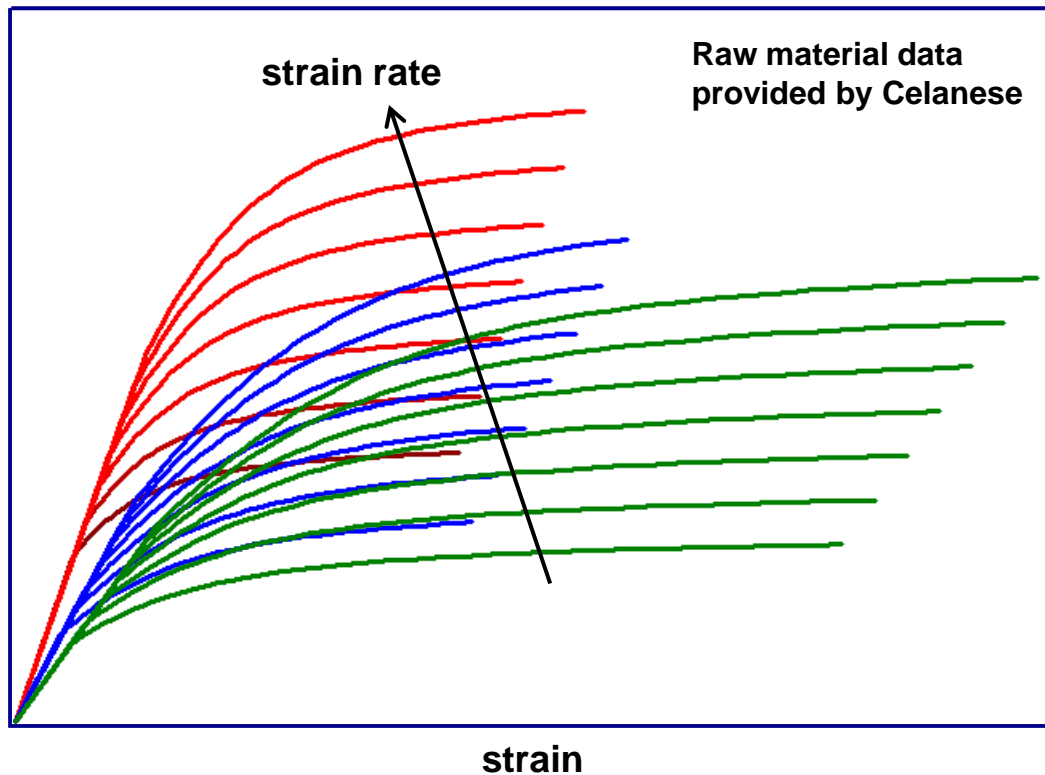


LS DYNA, ABAQUS, PAM CRASH, ...

Reverse Engineering

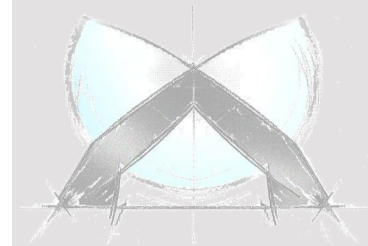
*MAT_24 – typical approach

Separate for each direction

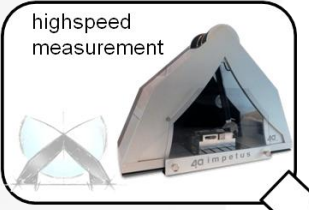


PP GF40

longitudinal
diagonal
perpendicular



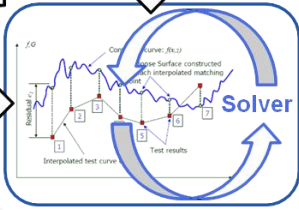
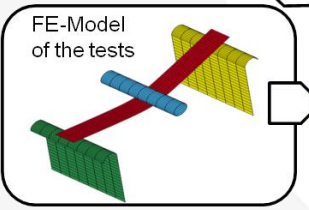
4a IMPETUS



parameterized materialcard

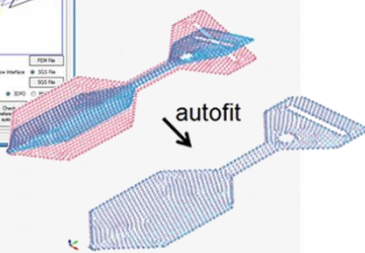
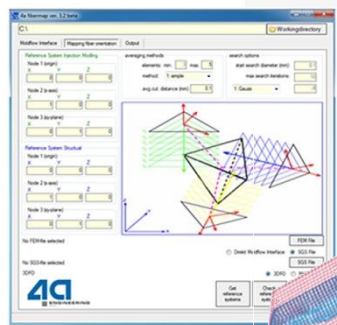
$$\sigma = \sigma_0 + E \cdot \varepsilon_p \cdot \frac{1}{\left[1 - \frac{E}{H} \cdot \varepsilon_p\right]}$$

$$\sigma = \sigma_0(\varepsilon) \left[1 + C \ln \frac{\dot{\varepsilon}}{\dot{\varepsilon}_0}\right]$$



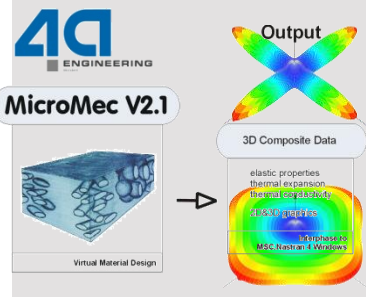
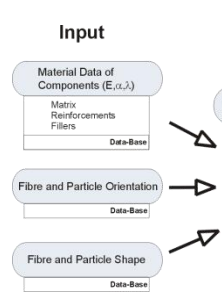
validated material cards

4a FIBERMAP

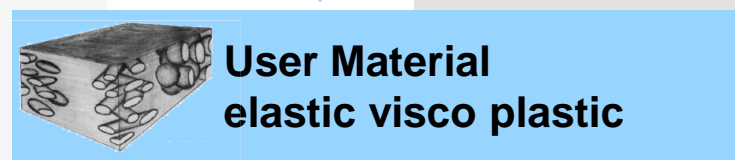
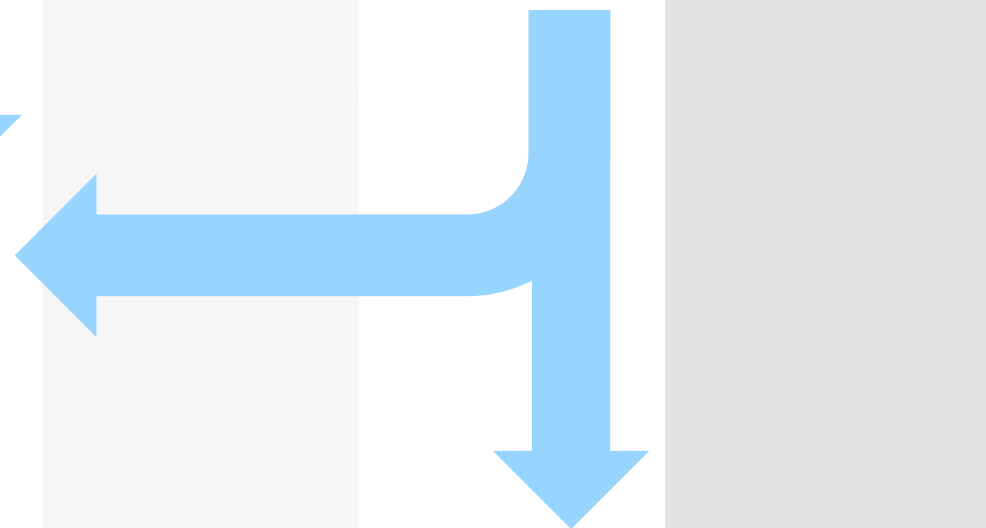
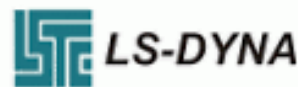
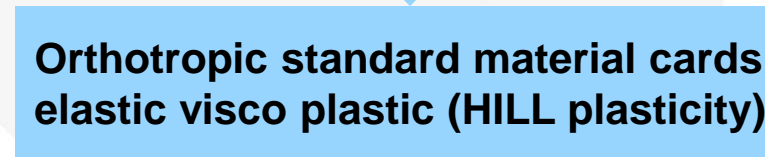
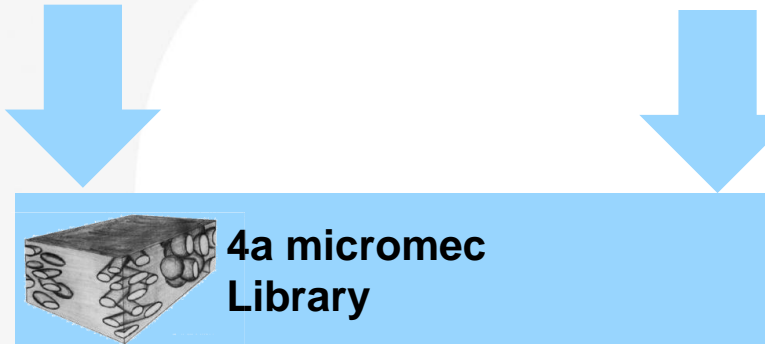


individual mapping process information

4a MICROMECH



3D thermo elastic anisotropic material cards



Micro mechanic based material models

Stand-alone product - calculating the elastic values

Input

Material Data of Components (E, α, λ)

Matrix
Reinforcements
Fillers

Data-Base



Fibre and Particle Orientation

Data-Base



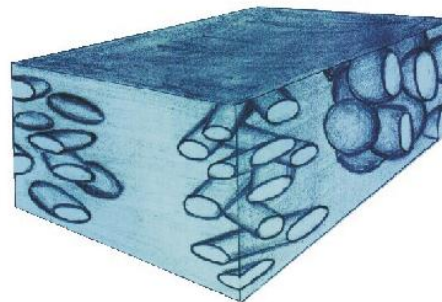
Fibre and Particle Shape

Data-Base



since 1999

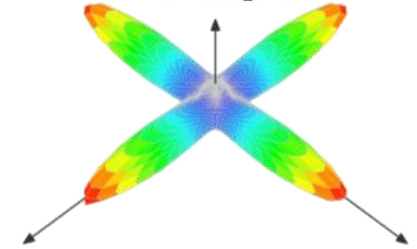
4a micromec



Virtual Material Design



Output



3D Composite Data

elastic properties
thermal expansion
thermal conductivity

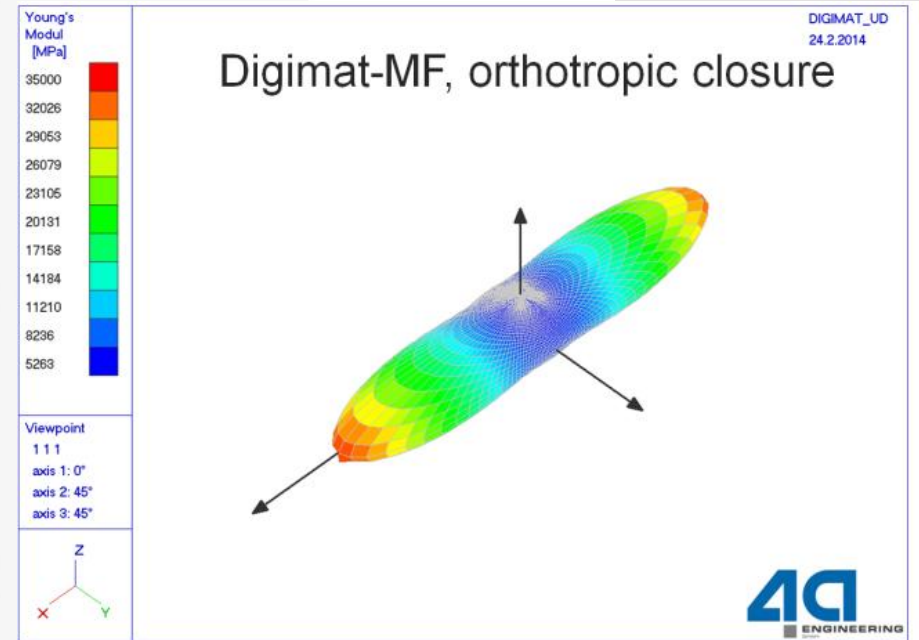
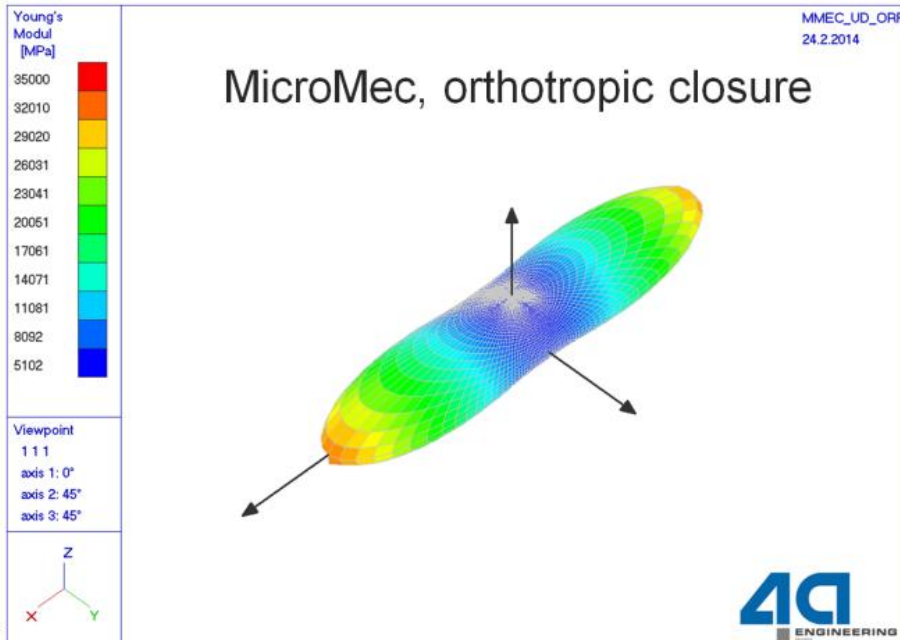
2D&3D graphics

Interphase to
MSC.Nastran 4 Windows

- SFRT, LFRT, CFRP, GFRP,
- 3D thermo-elastic properties

Micro mechanic based material models

Stand-alone product - calculating the elastic values



- Comparison by University of Leoben between Digimat-MF and 4a micromec
- User material (orthotropic elastic visco-plastic including failure)

No.	Elastic	Plastic	Damage	Strain rate	Failure	
2	Ortho / Anisotropic	None	None	None	*MAT_ADD_EROSION	SFRT / LFRT
24	Isotropic	Mises	None	Plasticity	*MAT_ADD_EROSION	
103	Isotropic	Hill	None	Plasticity	*MAT_ADD_EROSION	
108	Orthotropic	Hill	None	None	*MAT_ADD_EROSION	
157	Anisotropic	Hill	None	Plasticity	*MAT_ADD_EROSION	
215	*MAT_4a_micromec in development: Model based on MORITANAKA MEANFIELD					
22	Orthotropic	None	None	None	Orientation dependent	Carbon, Glass, Kevlar endless & fabric
54/55	Orthotropic	None	Elastic Orthotropic	Strength	Chang-Chang/ Tsai-Wu Orientation dependent	
58	Orthotropic	None	Elastic Orthotropic	Strength, Stiffness	mod. Hashin Orientation dependent	
158	Orthotropic	None	Elastic Orthotropic	Visco-elasticity	Orientation dependent	
261	Orthotropic	None	Elastic Orthotropic	None	failure Pinho (Puck) Orientation dependent	
262	Orthotropic	None	Elastic Orthotropic	None	failure Camanho (Puck) Orientation dependent	

USERMAT → Micro mechanic based material models

- ***MAT_157** → phenomenological model based on composites stresses
- ***MAT_4a_micromec (*MAT_215)**
 - Current development / available Q4/2016
 - Based on 4a_micromec
 - In each simulation cycle
 - **Composite stress → matrix and fiber stress**
 - Matrix: isotropic elastic viscoplastic
 - Fiber: (transversal) isotropic elastic
 - Advantage
 - Material behavior **local** defined **through mapped fiber orientation**
 - Failure criteria based on matrix or fiber

- Hill plasticity → „extended“ von Mises

$$\sigma_{\text{eq}} = \sqrt{F(\sigma_{22} - \sigma_{33})^2 + G(\sigma_{33} - \sigma_{11})^2 + H(\sigma_{11} - \sigma_{22})^2 + 2L\sigma_{23}^2 + 2M\sigma_{31}^2 + 2N\sigma_{12}^2}$$

$$F = \frac{(\sigma^0)^2}{2} \left(\frac{1}{\bar{\sigma}_{22}^2} + \frac{1}{\bar{\sigma}_{33}^2} - \frac{1}{\bar{\sigma}_{11}^2} \right) = \frac{1}{2} \left(\frac{1}{R_{22}^2} + \frac{1}{R_{33}^2} - \frac{1}{R_{11}^2} \right), \quad L = \frac{3}{2} \left(\frac{\tau^0}{\bar{\sigma}_{23}} \right)^2 = \frac{3}{2R_{23}^2},$$

$$G = \frac{(\sigma^0)^2}{2} \left(\frac{1}{\bar{\sigma}_{33}^2} + \frac{1}{\bar{\sigma}_{11}^2} - \frac{1}{\bar{\sigma}_{22}^2} \right) = \frac{1}{2} \left(\frac{1}{R_{33}^2} + \frac{1}{R_{11}^2} - \frac{1}{R_{22}^2} \right), \quad M = \frac{3}{2} \left(\frac{\tau^0}{\bar{\sigma}_{13}} \right)^2 = \frac{3}{2R_{13}^2},$$

$$H = \frac{(\sigma^0)^2}{2} \left(\frac{1}{\bar{\sigma}_{11}^2} + \frac{1}{\bar{\sigma}_{22}^2} - \frac{1}{\bar{\sigma}_{33}^2} \right) = \frac{1}{2} \left(\frac{1}{R_{11}^2} + \frac{1}{R_{22}^2} - \frac{1}{R_{33}^2} \right), \quad N = \frac{3}{2} \left(\frac{\tau^0}{\bar{\sigma}_{12}} \right)^2 = \frac{3}{2R_{12}^2},$$

Material models in LS-DYNA

*MAT_157 → *INITIAL_STRESS_SHELL(SOLID)

*MAT_157: Selective mapping IHIS

FLAG	Description	Variables	#
a_0	Material directions	$q_{11}, q_{12}, q_{13}, q_{31}, q_{32}, q_{33}$	6
a_1	Anisotropic stiffness	C_{ij}	21
a_2	Anisotropic constants	F, G, H, L, M, N	6
a_3	Stress-strain curve	$LCSS$	1

*INITIAL_STRESS_SOLID: NHISV

- in addition to 6 stress values and eps NHISV history can be initialized
- NHISV must correspond to the a_i that define IHIS in *MAT_157

$$NHISV = 6a_0 + 21a_1 + 6a_2 + 1a_3$$

Material models in LS-DYNA

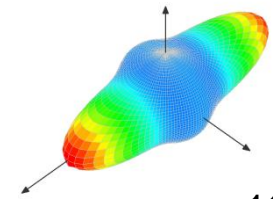
*MAT_157 material

- material model based on **composites properties**

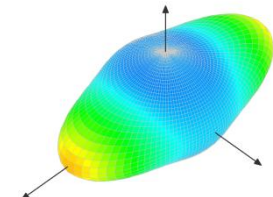
description	variables	Number of variables	dependencies
anisotropic stiffness	Cij	21	$C_{ij}(a_{ij}, \varphi, C^M, C^F)$
Hill plasticity	3D: F, G, H, L, M, N 2D: r00, r45, r90	6	$f(a_{ij}, \varphi, \sigma^M, \sigma^F)$
stress-strain curve	Loadcurve	3	$f(a_{ij}, \varphi)$
failure	Loadcurve	6	$f(a_{ij}, \varphi)$

- Not possible to generate samples with **explicit defined and varying a_{ij}**
- Hard to characterize, too many possibilities in a_{ij}

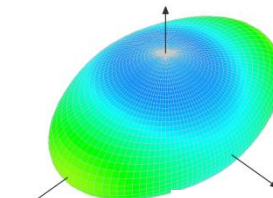
→ **Micro mechanical model is needed**



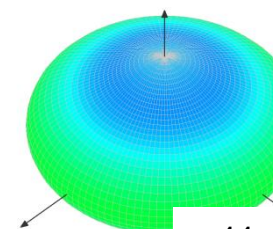
a11=1



a11=0.85



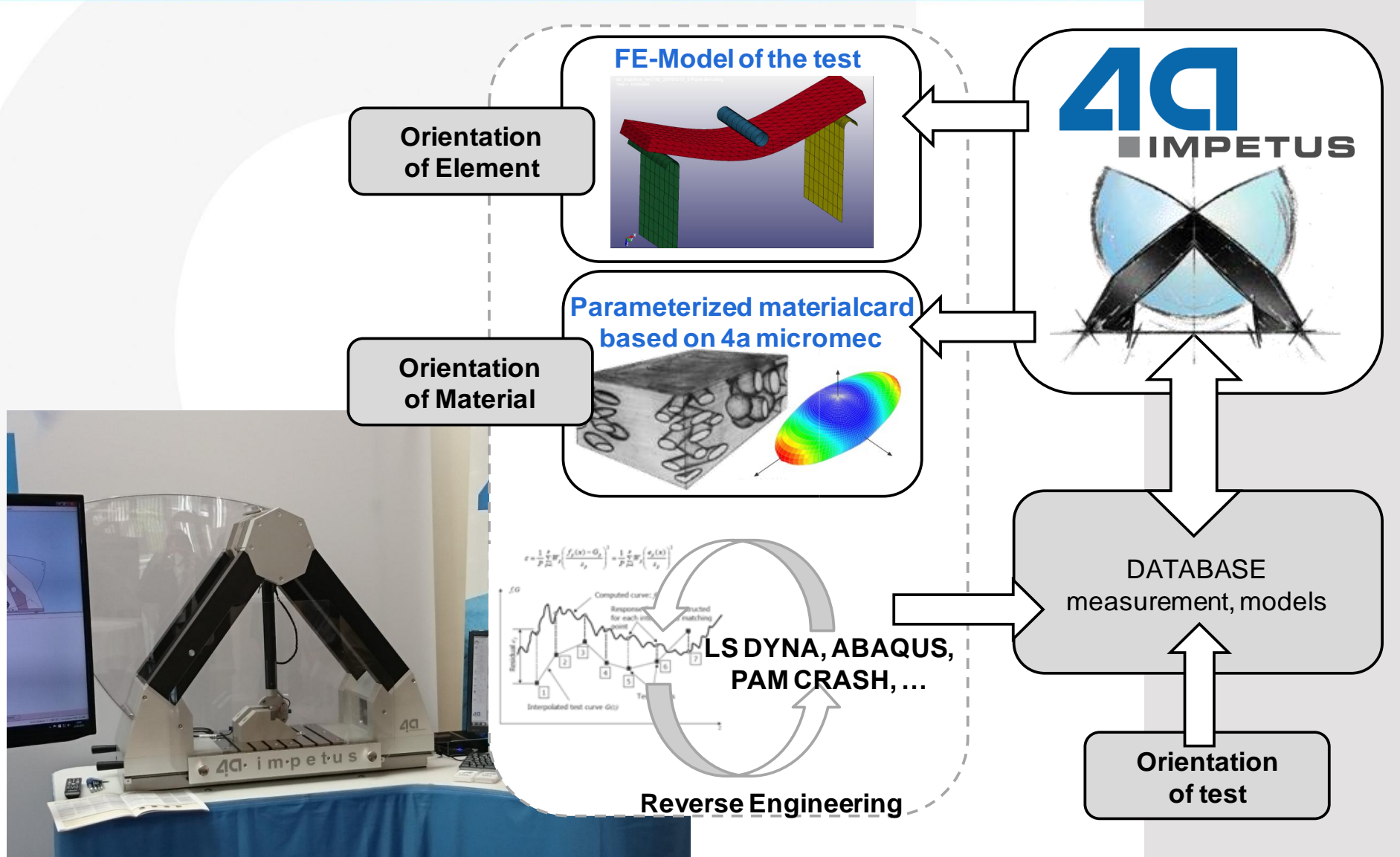
a11=0.7



a11=0.5

MAT_157 - micro mechanic models as key enabler

Consider each direction in one material model



Consider each direction in one material model

e.g. PP LGF30

- micro mechanics
based on **Fiber & Matrix** properties

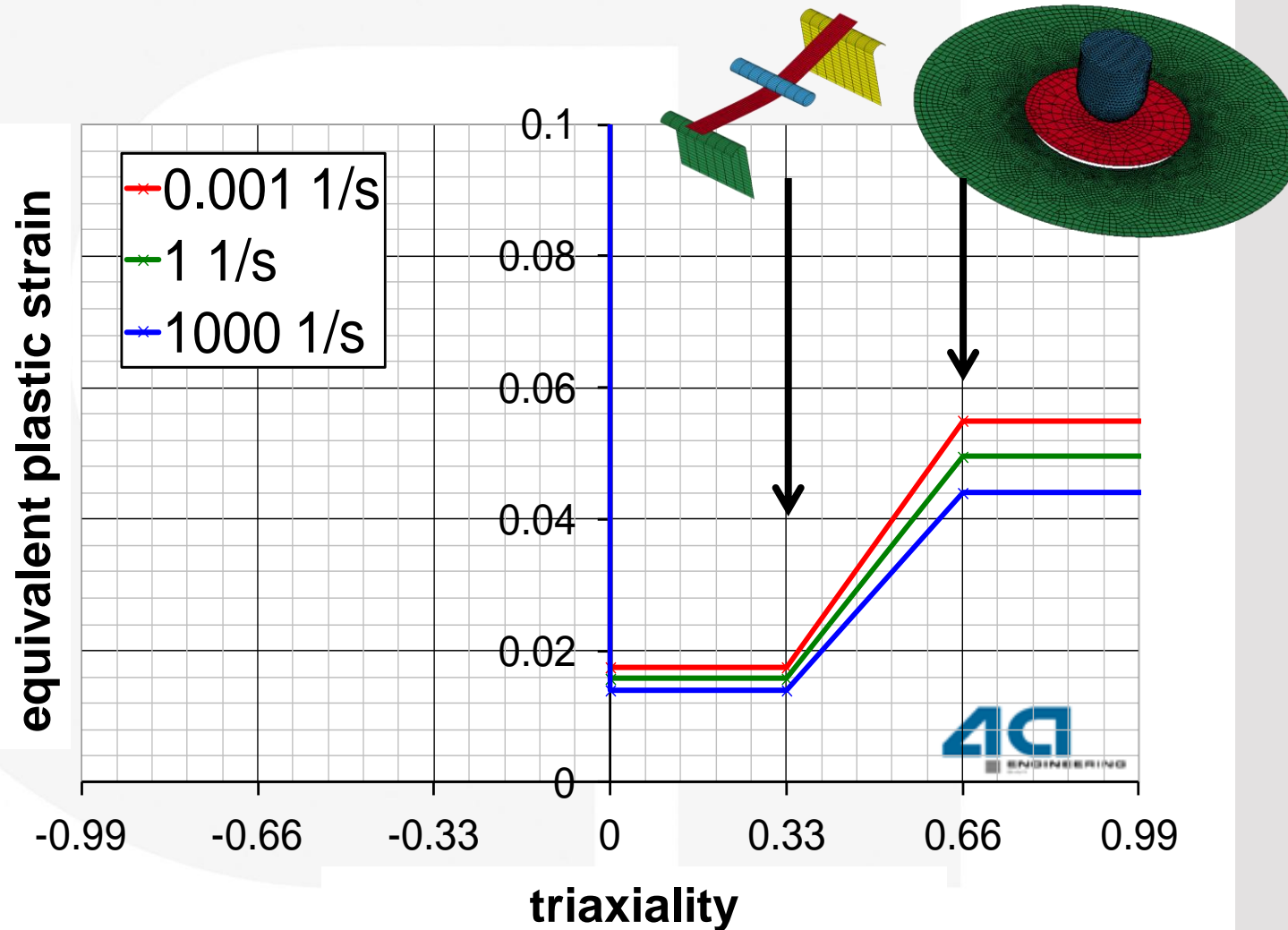
Composite elasticity → orthotropic $E_1, E_2, E_3, \nu_{12}, \nu_{13}, \nu_{23}, G_{12}, G_{13}, G_{23}$

Composite plasticity → HILL coefficient F, G, H, L, M, N

- **Reverse Engineering**

- hardening: Bilinear → **2 parameters**
- strainrate: Johnson Cook → **1 parameter**
- failure: DIEM → **equivalent HILL strain**

MAT_ADD_EROSION: DIEM with DUCTILE CRITERION



MAT_157 - micro mechanic models as key enabler

Current 4a impetus 3.4 alpha version

160223_006 Material Designvariablen Layers

<input type="checkbox"/> Strain rate dependency	Table
<input type="checkbox"/> Strain rate dependency	Johnson Cook
<input type="checkbox"/> Micromec	User defined
<input type="checkbox"/> Matrix	
Density of the matrix	900
E-Modulus	1500
Poisson's ratio	0.3
Yield strength	15
Strength at Break	17
Failure strain	0.05
<input type="checkbox"/> Fiber	
Fillerlength	1000
Fillerdiameter	20
Phi or Psi	φ
Phi	12.9
Psi	30.1
Fillermaterial	E-Glas
<input type="checkbox"/> Orientation	
<input type="checkbox"/> Fillerorientationtype	CA lin. OF
Fillerorientationvalue 1	0.6
Fillerorientationvalue 2	0.33

Composite Density	1126 [g/dm ³]
c_C11	6172 [MPa]
c_C12	1808 [MPa]
c_C13	1231 [MPa]
c_C14	0 [MPa]
c_C15	0 [MPa]
c_C16	0 [MPa]
c_C22	4135 [MPa]
c_C23	1181 [MPa]
c_C24	0 [MPa]
c_C25	0 [MPa]
c_C26	0 [MPa]
c_C33	2616 [MPa]
c_C34	0 [MPa]
c_C35	0 [MPa]
c_C36	0 [MPa]
c_C44	1554 [MPa]
c_C45	0 [MPa]
c_C46	0 [MPa]
c_C55	888.6 [MPa]
c_C56	0 [MPa]
c_C66	957.5 [MPa]
y_r00	1 [1]
y_r45	0.5105 [1]
y_r90	0.2665 [1]
y_scalematrix0	3.076 [1]

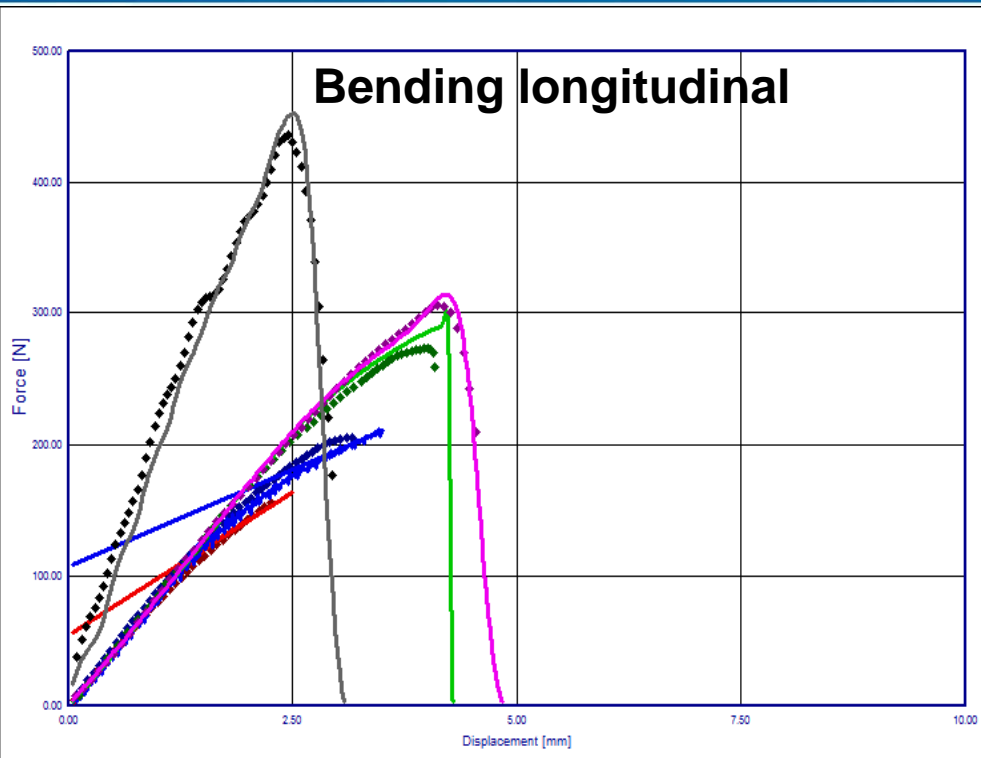
MAT_157 - micro mechanic models as key enabler

Current 4a impetus 3.4 alpha version

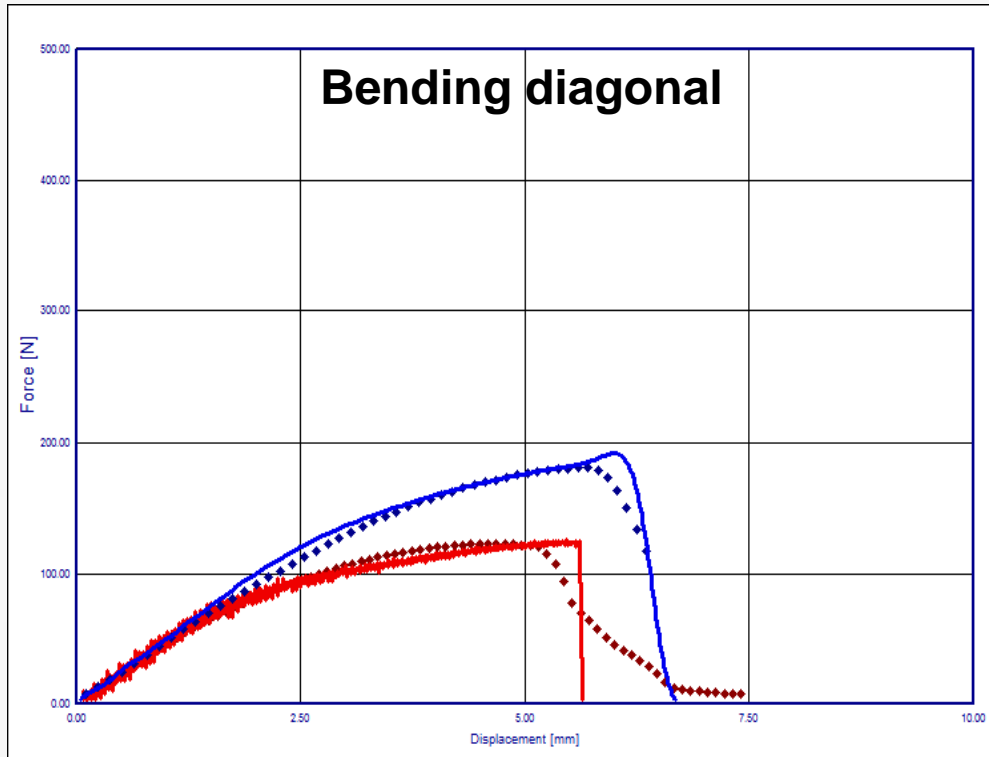
Name	Start	const...	Description
^ GroupName: 10_elasticity			
c_C11	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 11
c_C12	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 12
c_C13	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 13
c_C14	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 14
c_C15	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 15
c_C16	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 16
c_C22	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 23
c_C23	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 23
c_C24	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 24
c_C25	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 25
c_C26	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 26
c_C33	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 33
c_C34	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 34
c_C35	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 35
c_C36	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 36
c_C44	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 44
c_C45	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 45
c_C46	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 46
c_C55	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 55
c_C56	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 56
c_C66	MMEC	<input checked="" type="checkbox"/>	constitutive matrix 66

Name	Start	const...	Description
^ GroupName: 10_elasticity			
^ GroupName: 20_yield			
y_0	90	<input checked="" type="checkbox"/>	yield stress
y_scale...	MMEC	<input checked="" type="checkbox"/>	yield scale 11 direction
y_r00	MMEC	<input checked="" type="checkbox"/>	yield hill anisotropy ratio 0°
y_r45	MMEC	<input checked="" type="checkbox"/>	yield hill anisotropy ratio 45°
y_r90	MMEC	<input checked="" type="checkbox"/>	yield hill anisotropy ratio 90°
^ GroupName: 21_hardening			
h_ET	50	<input checked="" type="checkbox"/>	
h_y	90	<input checked="" type="checkbox"/>	
^ GroupName: 31_strainrate			
v_epspkt	0.01	<input checked="" type="checkbox"/>	initial strain rate threshold
v_p	15	<input checked="" type="checkbox"/>	strain rate scale (1/vp)
^ GroupName: 51_failure			
xf_NUM...	0.75	<input checked="" type="checkbox"/>	Number of failed integration points prior to

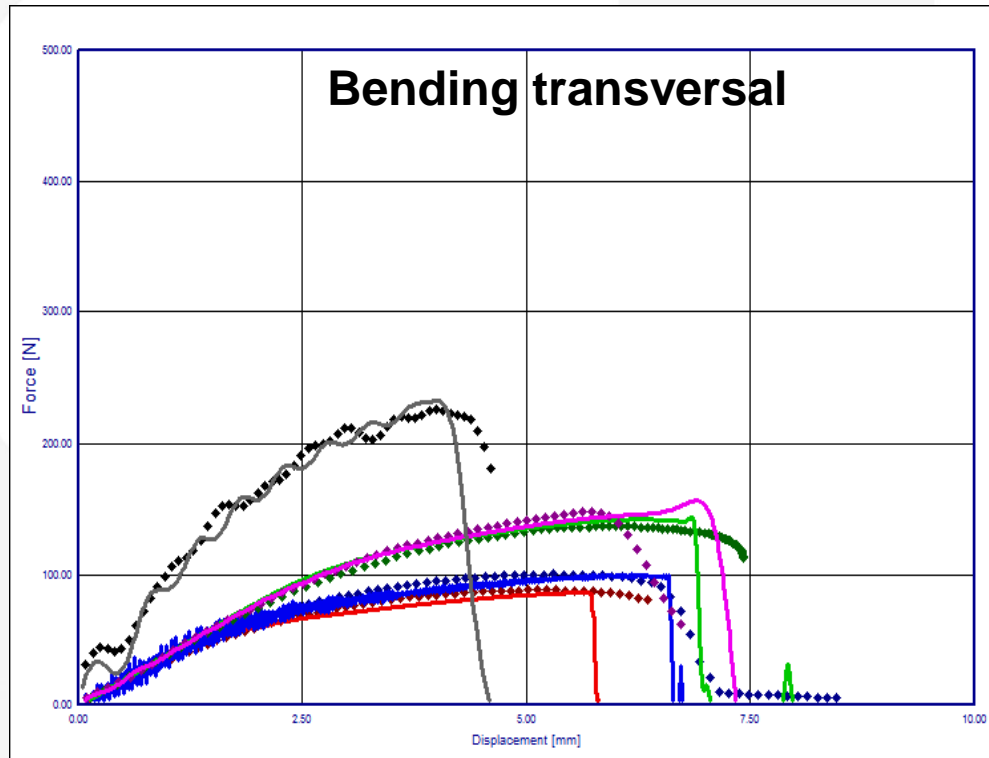
* [Click here to add a new row](#)



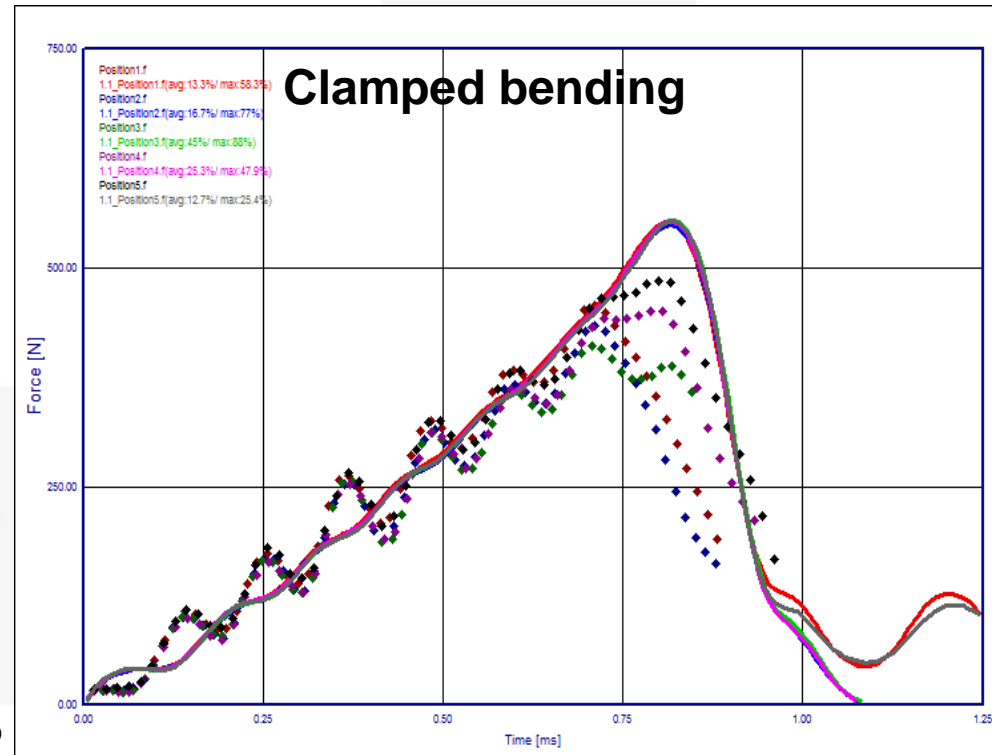
Shell 2mm, Type 16, 5 IP



Shell 2mm, Type 16, 5 IP



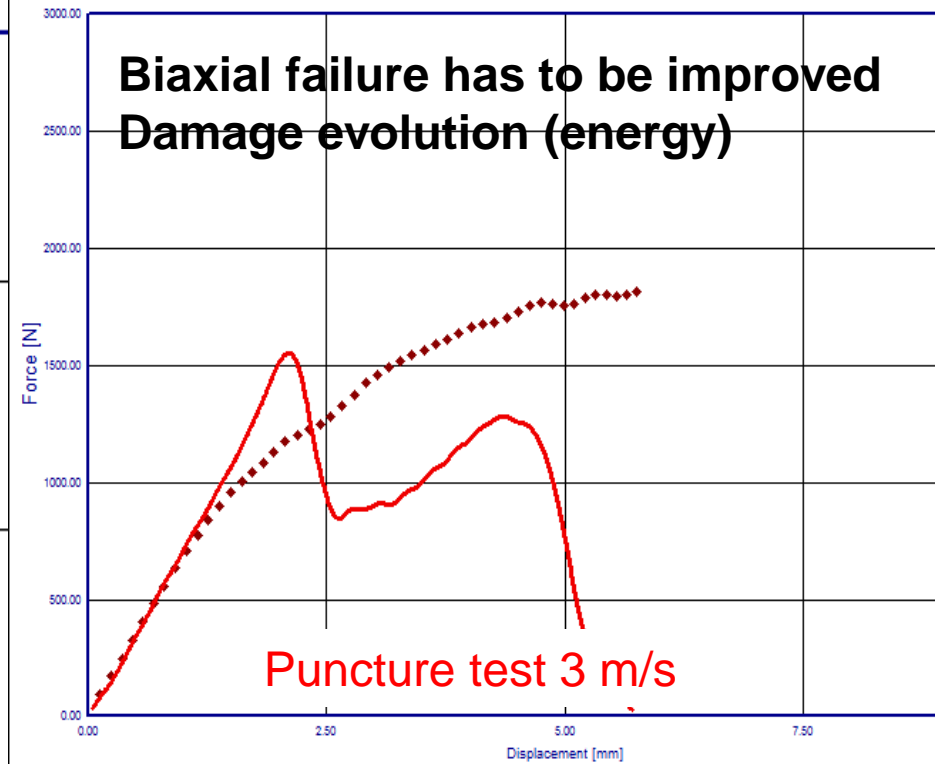
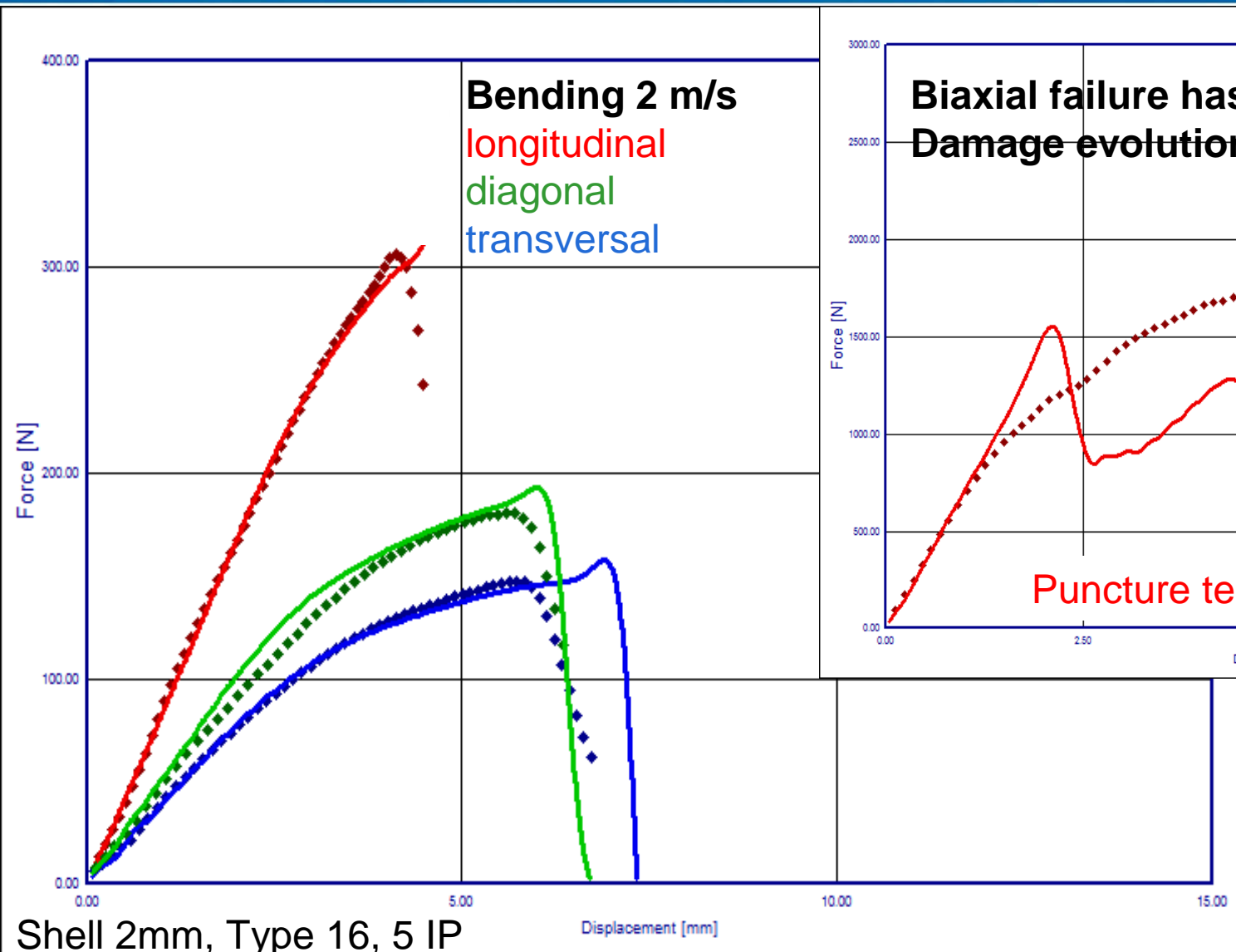
Shell 2mm, Type 16, 5 IP



Shell 2mm, Type 16, 5 IP

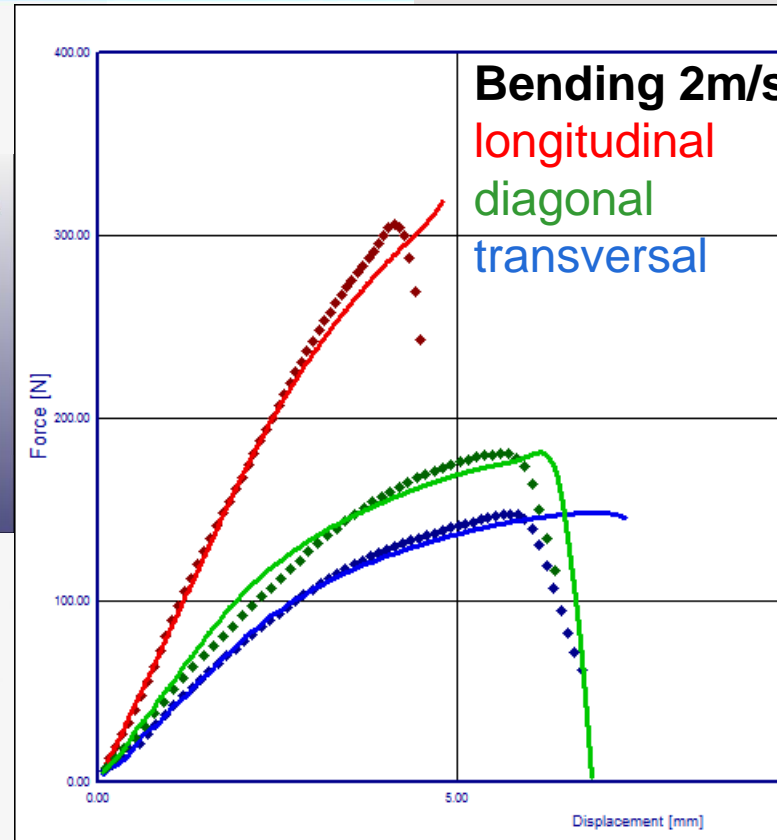
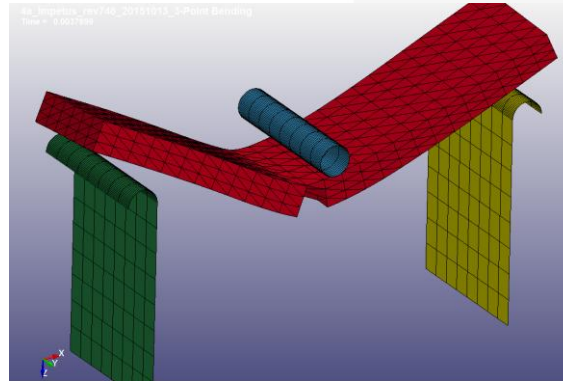
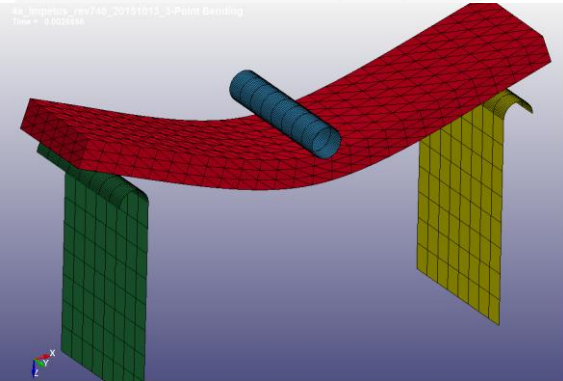
MAT_157 - micro mechanic models as key enabler

Consider each direction in one material model



Consider each direction in one material model

Solid – TET10 (Type 16)

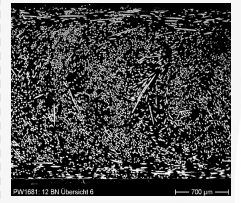


Simulation process chain

For injection molded parts

[2012Reithofer]

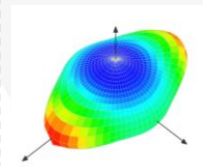
Injection molding (Moldflow, Moldex, ...)



$$a_{ij} = \begin{bmatrix} 0,66 & 0 & 0 \\ 0 & 0,32 & 0 \\ 0 & 0 & 0,02 \end{bmatrix}$$

fiber orientation

micromechanic modeling (4a micromec, ...)

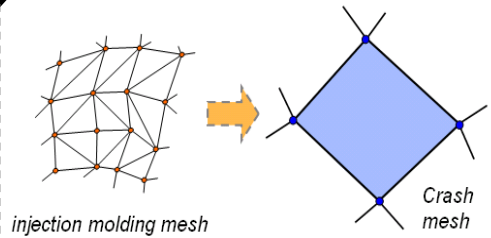


$$c = \begin{bmatrix} c_{11} & c_{12} & c_{13} & c_{14} & c_{15} & c_{16} \\ & c_{22} & c_{23} & c_{24} & c_{25} & c_{26} \\ & & c_{33} & c_{34} & c_{35} & c_{36} \\ & & & c_{44} & c_{45} & c_{46} \\ & & & & c_{55} & c_{56} \\ & & & & & c_{66} \end{bmatrix}$$

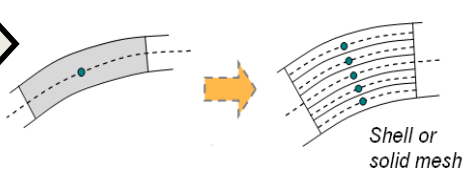
anisotropy

Mapping (4a fibermap, ...)

In plane mapping



Mapping in thickness direction



Crash Simulation (LS-DYNA)

orientations

- *ELEMENT_SHELL_BETA
- *ELEMENT_SHELL_COMPOSITE
- *INITIAL_STRESS_SHELL(SOLID)
- *ELEMENT_SOLID_ORTHO

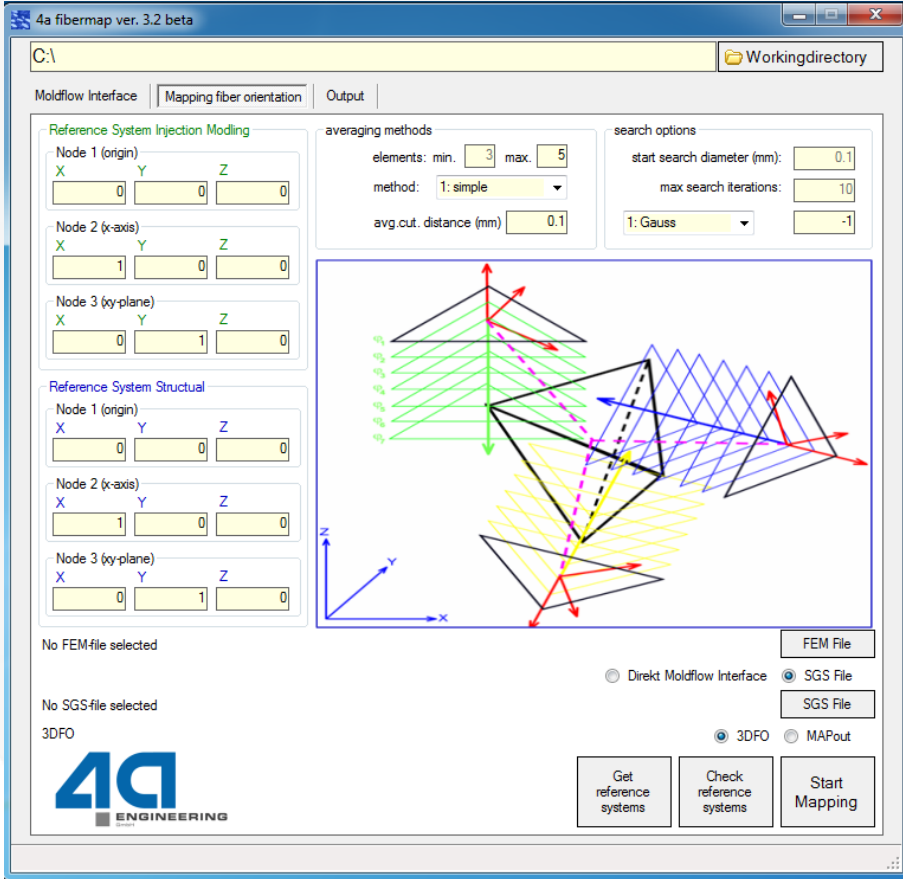
standard material models

- *MAT_(ANISO)TROPIC_ELASTIC
- *MAT_ANISOTROPIC_ELASTIC_PLASTIC



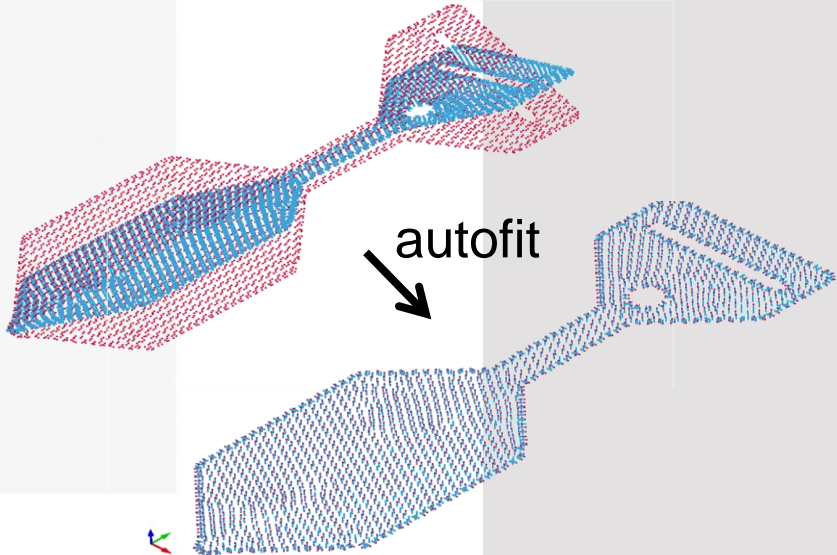
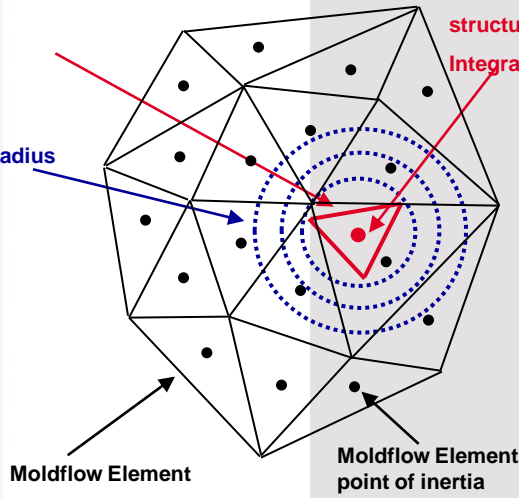
Simulation process chain

4a fibermap - mapping

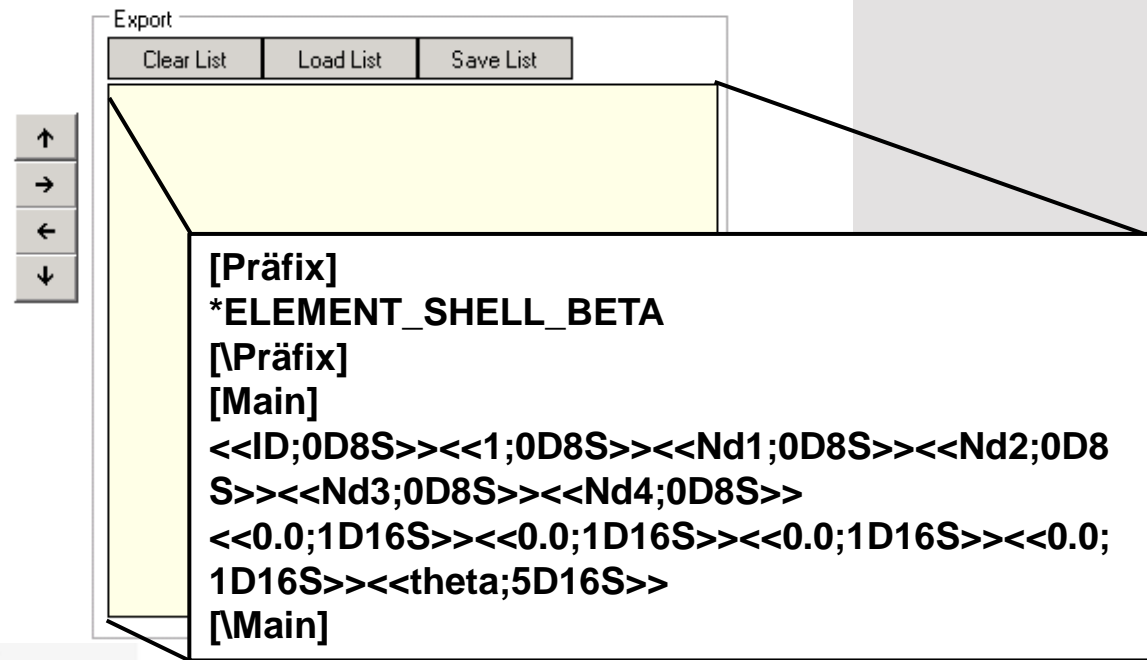
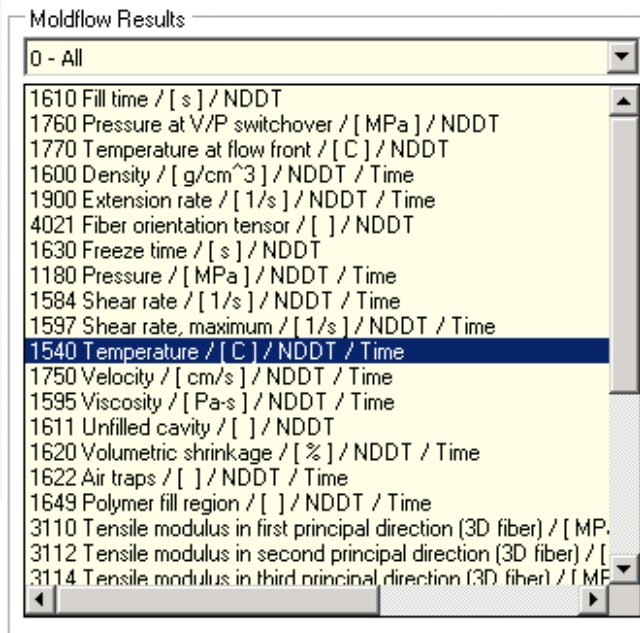


structural mesh element
structural mesh integration point

Search radius



- Moldex3D®- fiber orientation
- Autodesk Moldflow® - fiber orientation and further results can be mapped

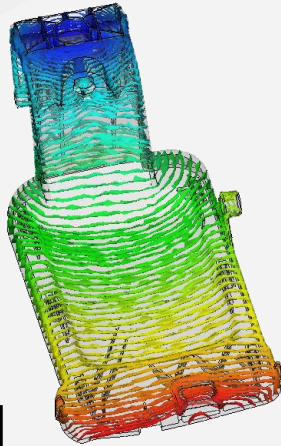


[2012Gramling]

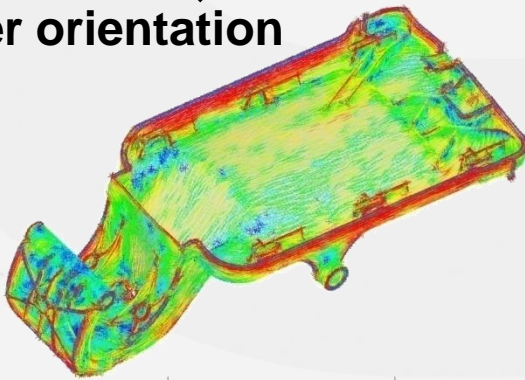
Process simulation

Filling

Fill time
= 1.736[s]

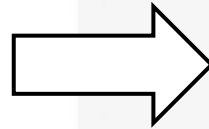
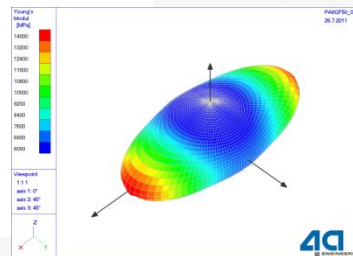


Fiber orientation



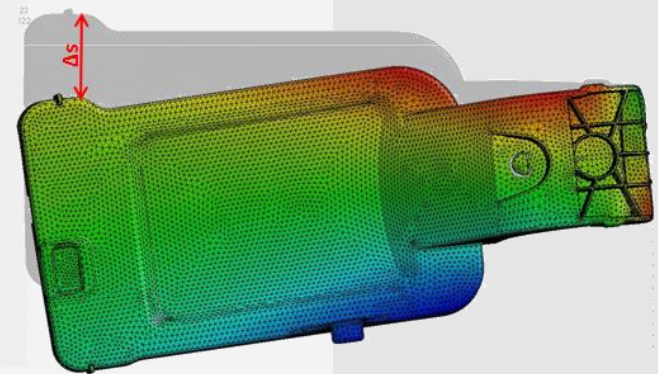
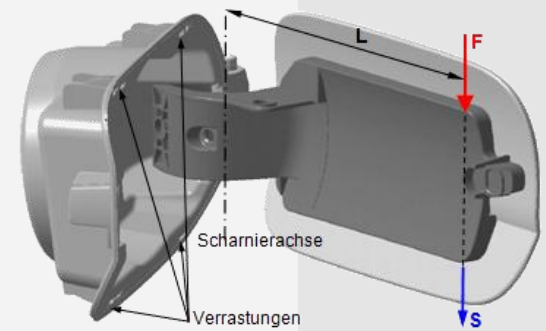
Scale (100 mm)

Mapping



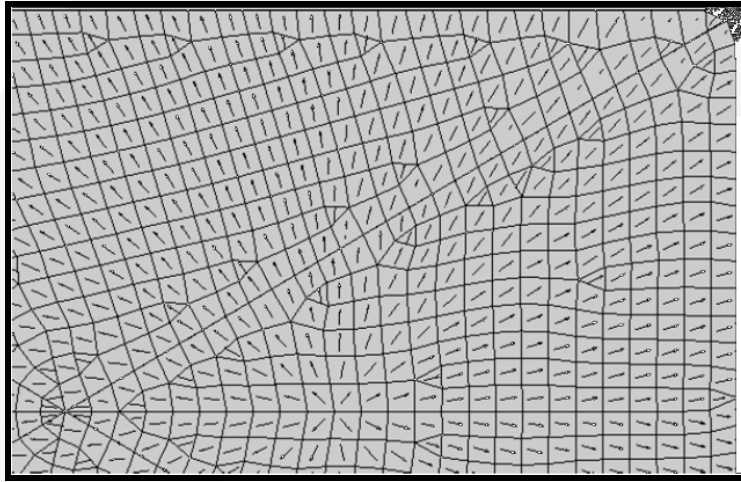
local anisotropy

Structural simulation bending

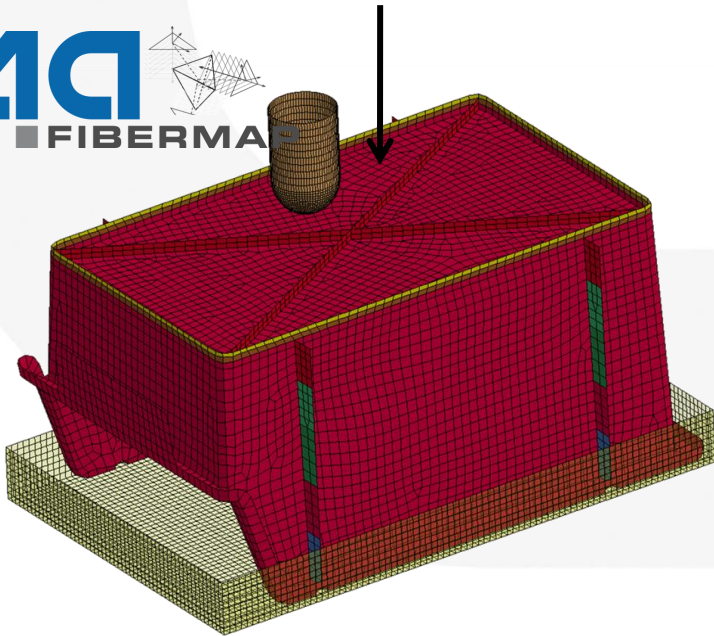
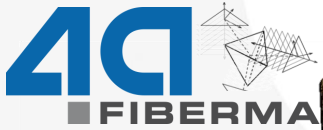


Case study Nutini Box

Using *MAT_157

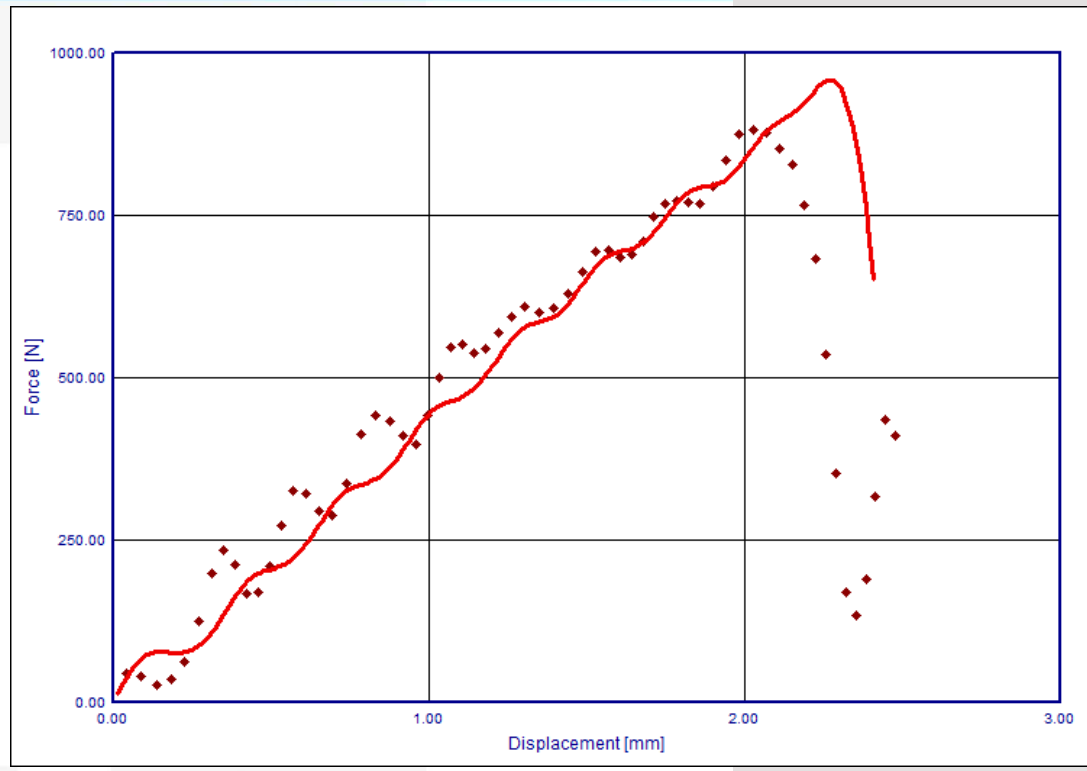


[2014Jennrich]

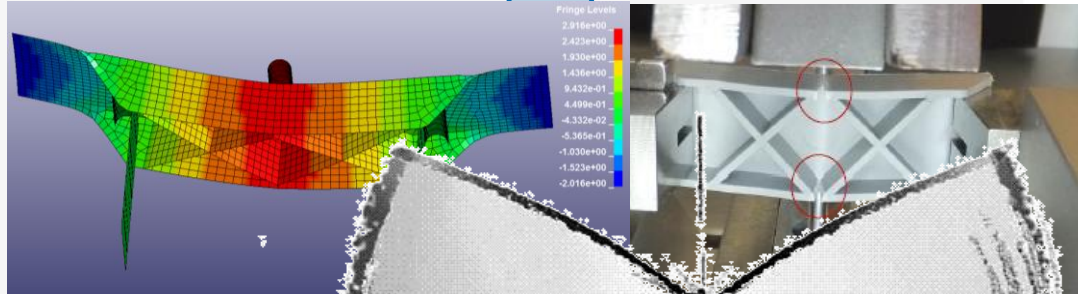


Case study Doublecrossrib

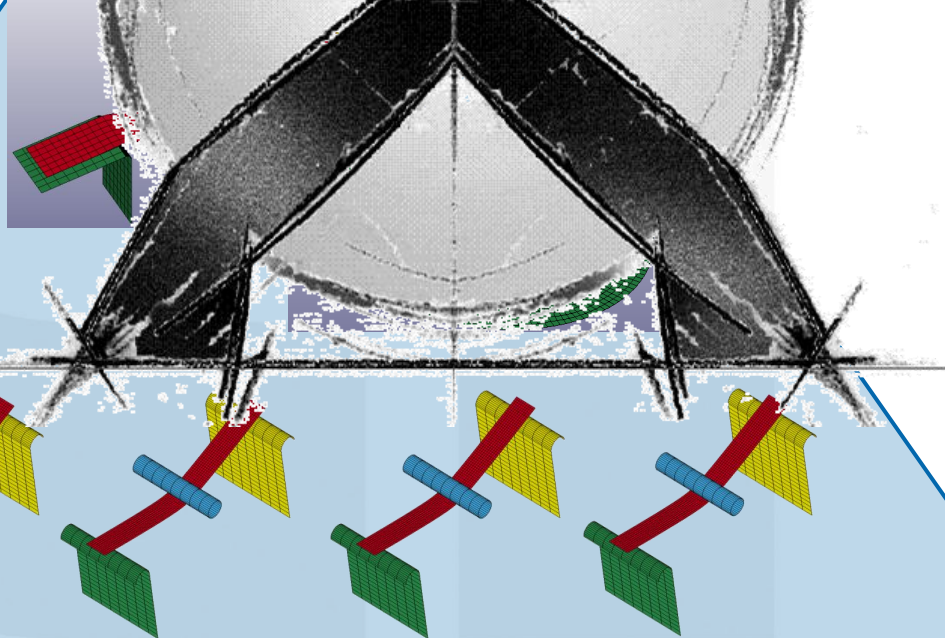
Using *MAT_215



Conclusion characterization pyramid micro mechanic models as key enabler

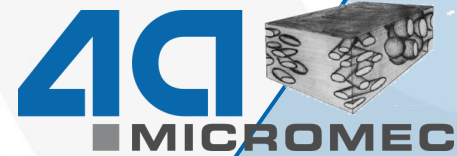


**component
validation**



failure

**elasticity
plasticity
strainrate**



Thank you for your attention!



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- [2008Reithofer]** P. Reithofer, M. Fritz, T. Wimmer (4a engineering GmbH) – *Kurzfaserverstärkte Kunststoffbauteile - Einfluss der prozessbedingten Faserorientierung auf die Strukturmechanik*, 7. LS-DYNA Anwenderforum, Bamberg 2008 ([Link](#))
- [2012Gramling]** M. Gramling (Audi AG), P. Reithofer (4a engineering GmbH) – *Integrative Simulation von kurzfaserverstärkten Thermoplasten am Beispiel einer Tankklappe*, 4a Technologietag, Schladming 2012 ([Link](#))
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- [2014Haufe]** A. Haufe (DYNAMore GmbH) – *Zum aktuellen Stand der Simulation von Kunststoffen mit LS-DYNA*, 11. 4a Technologietag - 2014 ([Link](#))
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- [2014SimuliaManual]** Abaqus Analysis User's Guide 6.14, Simulia - Dassault Systèmes, <http://simulia.custhelp.com>